RESEARCH ARTICLE



Design and validation of a novel adaptive motion control for a pendulum spherical robot

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Received: 1 April 2022; Revised: 20 January 2023; Accepted: 17 February 2023; First published online: 16 March 2023

Keywords: spherical robot, dynamic model, state space, motion controller, self-adaptation, Lyapunov method

Abstract

Spherical robots (SRs) have the characteristics of nonholonomic constraints, underactuation, nonchain, and strong coupling, which increase the difficulty of modeling and motion control compared with traditional robots. In this study, we develop an adaptive motion control scheme for a nonholonomic SR, in which an omnidirectional dynamic model is carried out by using the Euler–Lagrange method to describe the omnidirectional motion of the SR more accurately. Furthermore, to facilitate the design of the motion controller, the dynamic model is simplified to obtain the state space expression of the SR. Aiming at the problem of poor control effect caused by the change of system model parameters which are influenced by dynamic model reduction, an adaptive motion control law of SR is designed based on MRAC. And the coefficient adjustment of the controller is obtained by the Lyapunov method, with the guaranteed stability of the closed-loop system. Finally, the controller designed in this thesis is compared with four controllers including linear quadratic regulator, Fuzzy PID, PSO-ADRC, and hierarchical SMC. The experimental comparison proves that the control scheme proposed in this study still has good control ability when the motion parameters are disturbed.

1. Introduction

Since the appearance of spherical robots (SR) in 1996, they have received extensive attention from scholars at home and abroad due to their compact structure, flexible movement, and ability to withstand various task loads [1]. The SR is a fully enclosed robot in which the drive mechanism, control system, and battery are all enclosed in the spherical shell. This prevents adverse damage to the external environment, which makes the SR suitable to be applied in humid, dusty, radiation, or toxic environment.

At present, the drive mechanism of the SR is mainly divided into omnidirectional wheel eccentric torque drive [2], frictional internal drive [3], and eccentric torque drive of heavy pendulum [4]. The pendulum drive is favored by industry and academia due to its simple implementation and significant driving torque. However, during the pendulum SR movement, the internal suspension is in the under actuation state, which affects its stability and accuracy, and even causes control failure.

In recent years, most studies on the pendulum SR's dynamic modeling and motion control have mainly focused on the speed control of the SR. For instance, a motion control approach was proposed in ref. [5] for a spherical mobile robot rolling on a plane, in which only one Euler angle was varied in each step, improving computational efficiency. In refs. [6–8], the motion equation was decomposed into horizontal and vertical subsystems. References [6, 7] designed full-state feedback and PI controllers for rolling and forward motions, respectively. In ref. [8], a feedback linearized PD-type fuzzy controller was employed for the closed-loop control of robot motion. Roozegar M. et al. [9] utilized the heuristic fuzzy controller and PID controllers to realize the SR's motion control on an inclined plane. The above approaches did not consider the system's robustness to parametric uncertainties and the reduction of stationary errors in detail. Wang Y. et al. [10] constructed a yaw angle prediction-based fuzzy PID

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controller to adjust the attitude of a SR, which can perform fuzzy reasoning and autonomously modify control parameters based on the robot's current state information. However, the controller is based on a kinematics design, which cannot effectively control the SR under the uncertain parameters of its model. Xie L. et al. [11] studied the dynamics modeling and joint trajectory tracking control of robotic systems with external disturbances and uncertain parameters and proposed a robust fuzzy sliding mode control to achieve desired trajectory tracking of joints. The synovial controller is independent of object model parameters and disturbances, the sliding modes of which can be designed.

In order to solve the problems of inaccurate dynamic modeling, parameter changes, and interference in the SR control that led to poor performance of traditional controllers, Cai Y. et al. [12] proposed a speed control method combining fuzzy logic and SMC, while Kayacan E. et al. [13] utilized the adaptive neuro-fuzzy controller and the learning algorithm of the SMC theory for the SR's speed control. Liu Y. et al. [14, 15] designed a hierarchical SMC (HSMC) for speed and orientation control of the SR, improving control efficiency and stability. Nevertheless, it is challenging to fit sensors in the limited space of the sphere to measure variables such as rolling speed in spherical robots. Therefore, applying the HSMC method for the motion control of SRs is challenging due to the unmeasurable state and the uncertain rolling resistance torque generated by the ground. Yue M. et al. [16] combined HSMC and the state expansion observer to promote the designed closed-loop system's robustness and adaptability. As we all know, since the state observer can reconstruct the system with the measurable state and obtain the value of the internal state variable, the uncertain rolling resistance can be obtained in real time.

Most of the above studies are in the simulation step, and even if there are physical experiments, they are mostly confined to smooth ground applications. SR control systems are susceptible to unmodeled dynamics, parameter changes, uncertainties, and disturbances. For example, the parameters are susceptible to disturbance when the spherical rolling robot rolls on uneven ground. Active disturbance rejection controller (ADRC) [17] is independent of the system model and can estimate and compensate for different disturbances to the system in real time during motion. Tao T. et al. [18] and Lin R. et al. [19] proposed an ADRC-based motion controller for SRs moving on uneven ground, which enhanced their anti-interference capability. Liu M. et al. [20] designed a motion controller (PSO-ADRC) that employs particle swarm optimization (PSO) to adjust the parameters b_0 , β_{01} , β_{02} , and β_{03} in ADRC to enhance the SRs' anti-interference efficiency in actual motion. However, the ADRC controller has many parameters where their tuning can considerably influence the SR's motion control efficiency.

There are other concerns about improving the control capability of disturbed parameters during the motion of SRs. For example, Yue M. [21], Sun H. [22], and Loh R.N.K. [23] designed the optimal control law of a linear quadratic regulator (LQR) with the change of state variables as the objective function to reduce the disturbance caused by the dynamic change of the state effects on SRs. Zadeh F. K. et al. [24] designed an efficient LQR to eliminate the oscillation in a SR's motion and improve the system's stability. Zhan Q. et al. [25–29] utilized backstepping, the neurodynamic shunt model, the cerebellar model joint control model, and RBF neural network to improve the controller's disturbance immunity.

Although the above control methods ensure the SR's stable control to a certain extent, the motion parameters of SRs vary significantly due to the lack of their environmental perception ability and susceptibility to external and internal uncertainties. Therefore, designing the control system based on an accurate model has certain limitations, especially when the system is greatly disturbed, degrading the system's control effect significantly. Model reference adaptive control (MRAC) is a powerful tool for solving the mentioned problem [30]. Typically, an MRAC comprises two distinct loops, where the inner one, including the plant and the controller, is connected in parallel with the reference model, and the outer one is an adaptation mechanism that adjusts the controller parameters. Since the reference signal applied to the adjustable control effect due to the parameter change of the system model so that the system can obtain better control performance. For example, Roozegar M. et al. [31] developed an MRAC-based adaptive method for the control of pendulum-driven SRs. Bomfim M. et al. [32] designed a Lyapunov-based hybrid MRAC for a pneumatic artificial muscle-actuated manipulator.



Figure 1. XK-I spherical robot coordinate system.

Inspired by the above research, a Lyapunov-based MRAC control algorithm is proposed. The essential contributions of the current work are (1) According to the actual situation of SR motion, taking the pendulum's swing angle and the spherical shell's rolling angle as generalized coordinates, the Euler– Lagrange approach is adopted to establish the SR learning model's omnidirectional dynamics. (2) The dynamic model is simplified sequentially to design the controller. According to the steering principle of the car during driving, the XK-I SR's motion control model is decoupled into linear and steering, and the state space expressions of the SR's linear and steering motions are obtained. (3) The proposed MRAC control algorithm is based on the Lyapunov method. MRAC is utilized to design the SR's adaptive motion control law, and the controller parameters are adjusted through the Lyapunov approach to overcome the poor control effect induced by the SR's inaccurate dynamic model and parameter uncertainty. Finally, the paper conducts experimental verification and compares MRAC with LQR, Fuzzy-PID, PSO-ADRC, and HSMC.

The remainder of this article is arranged as follows. Section 2 describes the XK-I SR's mathematical model, including the uncoupled dynamic model and state space equations. Section 3 designs the adaptive motion controller for the SR. The simulation and experimental results are given in Section 4. In the end, the conclusions of this study and future research perspectives are presented in Section 5.

2. Description of the pendulum SR system

2.1. Dynamic model of the pendulum SR

The XK-I SR in this study is composed of a two-degree-of-freedom pendulum driven by two sets of motors. As described in Fig. 1, a Cartesian coordinate system $o - x_1y_1z_1$ is established, considering the sphere's center as the origin o. The driving torque produced by the motor makes the pendulum swing around the ox_1 and oy_1 axes by angles α and β , respectively. In Fig. 1, o - xyz describes an inertial coordinate system fixed on the moving ground, and $o - x_0y_0z_0$ describes a moving coordinate system with the center of the sphere as the origin. The $o - x_0y_0z_0$ coordinate system is only translated relative to the $o - x_0y_0z_0$ coordinate system, and the $o - x_1y_1z_1$ coordinate system is only rotated with respect to the $o - x_0y_0z_0$ coordinate system. θ , φ , and ψ are the rolling angles of the sphere around ox, oy, and oz, respectively. From the actual motion of the XK-I type, only considering the sphere rotation around the lateral axis ox and the longitudinal axis oy, the rotation matrix between $o - x_1y_1z_1$ and $o - x_0y_0z_0$ is

$$\boldsymbol{R} = \boldsymbol{R}(x,\theta)\boldsymbol{R}(y,\varphi) = \begin{bmatrix} \cos\varphi & 0 & -\sin\varphi \\ \sin\theta\sin\varphi & \cos\theta & \sin\theta\cos\varphi \\ \cos\theta\sin\varphi & -\sin\theta & \cos\theta\cos\varphi \end{bmatrix}$$
(1)

In order to construct the XK-I SR's kinematics and dynamics models, the following assumptions are made:

- 1. There is no sliding friction when the ball moves on the ground;
- 2. The spherical shell's density and thickness are uniformly distributed, and the mass is m_1 ;
- 3. The square frame of the XK-I type is like a pendulum, and the mass is m_2 ;
- 4. The pendulum's center is connected to the sphere's center with a rigid rod regardless of mass, and the length is *l*;
- 5. The whole system's center of mass is located at the spherical shell's geometric center;
- 6. The sphere's rotational motion around the vertical oz axis is ignored;
- 7. The swing direction is specified as the positive direction.

Assuming the spherical shell center of mass's angular velocity is $\boldsymbol{\omega}_1 = (-\dot{\theta}, -\dot{\varphi}, 0)^T$, then the corresponding velocity is $\boldsymbol{v}_1 = r\boldsymbol{\omega}_1 = (-r\dot{\theta}, -r\dot{\varphi}, 0)^T$, where *r* is sphere radius. Thus, the spherical shell's Lagrangian function is

$$L_1 = E_1 - P_1$$
 (2)

where E_1 describes the sum of the spherical shell's horizontal displacement and rotational kinetic energies. Since the spherical shell is equivalent to a uniform thin spherical shell, its moment of inertia is $I_1 = \frac{2}{3}m_1r^2$. Thereby,

$$E_{1} = \frac{1}{2}m_{1} \|\boldsymbol{v}_{1}\|^{2} + \frac{1}{2}I_{1} \|\boldsymbol{\omega}_{1}\|^{2}$$
(3)

where P_1 is the gravitational potential energy of the spherical shell, and the rolling plane is a zeropotential energy surface, then: $P_1 = m_1 gr$.

Assuming that the pendulum's angular velocities in the $o - x_0y_0z_0$ and $o - x_1y_1z_1$ coordinate systems are denoted by ω_2 and ω , respectively, then,

$$\boldsymbol{\omega}_2 = \boldsymbol{\omega}_1 + \boldsymbol{R}\boldsymbol{\omega} = (\dot{\alpha}\cos\varphi - \dot{\theta}, \dot{\alpha}\sin\theta\sin\varphi + \dot{\beta}\cos\theta - \dot{\varphi}, \dot{\alpha}\cos\theta\sin\varphi - \dot{\beta}\sin\theta)^T$$
(4)

where $\boldsymbol{\omega} = (\dot{\alpha}, \dot{\beta}, 0)^T$. The position vector of the pendulum's center of mass is described as \boldsymbol{r}_2 and \boldsymbol{r} in the $o - x_0 y_0 z_0$ and $o - x_1 y_1 z_1$ coordinate systems, respectively, then

$$\boldsymbol{r}_2 = \boldsymbol{R}\boldsymbol{r} \tag{5}$$

Since the pendulum can swing around the ox_1 and oy_1 axes, the position of the pendulum's center of mass changes in the three-dimensional space, then

$$\mathbf{r} = x\mathbf{i} + y\mathbf{j} + z\mathbf{k} = (-l\sin\beta)\mathbf{i} + (l\sin\alpha)\mathbf{j} - (l\cos\alpha\cos\beta)\mathbf{k}$$
(6)

where (i, j, k) describes a unit vector. The linear velocity of the pendulum's center of mass is

$$\boldsymbol{v}_2 = \boldsymbol{v}_1 + \boldsymbol{\omega}_2 \times \boldsymbol{r}_2 \tag{7}$$

The Lagrangian function of the pendulum is $L_2 = E_2 - P_2$, where E_2 is the sum of the pendulum's horizontal movement and rotational kinetic energies, which is

$$E_2 = \frac{1}{2}m_2 \|\boldsymbol{\nu}_2\|^2 + \frac{1}{2}I_2 \|\boldsymbol{\omega}_2\|^2$$
(8)

where I_2 is the pendulum's moment of inertia, $I_2 = \frac{1}{12}m_2l^2 + \frac{1}{4}m_2l^2$. P_2 describes the sum of the pendulum's potential energy, $P_3 = m_2g(r - l\cos\alpha\cos\beta)$. Combining Eqs. (1)–(8), the whole XK-1 SR system's Lagrangian function is attained as

$$L = \frac{1}{2}m_1 \|\mathbf{v}_1\|^2 + \frac{1}{2}I_1 \|\boldsymbol{\omega}_1\|^2 - m_1gr + \frac{1}{2}m_2 \|\mathbf{v}_2\|^2 + \frac{1}{2}I_2 \|\boldsymbol{\omega}_2\|^2 - m_2g \left(r - l\cos\alpha\cos\beta\right)$$
(9)

From the Lagrangian approach, the pendulum swing angles α , β and spherical shell rolling angles θ , φ are taken as four generalized coordinates, that is, $\boldsymbol{q} = (\theta, \alpha, \varphi, \beta)^T$. The SR movement is generally driven by the two motors torques, τ_x and τ_y output by the two sets of motors. Therefore, the input torque of the system is $\boldsymbol{Q} = (\tau_x, \tau_x, \tau_y, \tau_y)^T$. Now, the motion's Euler–Lagrangian equation for the simplified model is attained as

$$\begin{cases} \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_1} \right) - \frac{\partial L}{\partial q_1} = \tau_x \\ \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_2} \right) - \frac{\partial L}{\partial q_2} = \tau_x \\ \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_3} \right) - \frac{\partial L}{\partial q_3} = \tau_y \\ \frac{d}{dt} \left(\frac{\partial L}{\partial \dot{q}_4} \right) - \frac{\partial L}{\partial q_4} = \tau_y \end{cases}$$
(10)

According to Eq. (10), the SR's dynamic equation is described with the following matrix form:

$$M(q)\ddot{q} + V(q,\dot{q})\dot{q} = u \tag{11}$$

$$\begin{pmatrix} m_{1} & a & b & c \\ a & m_{2} & d & e \\ b & d & m_{3} & f \\ c & e & f & m_{4} \end{pmatrix} \cdot \begin{bmatrix} \ddot{q}_{1} \\ \ddot{q}_{2} \\ \ddot{q}_{3} \\ \ddot{q}_{4} \end{bmatrix} + \begin{bmatrix} V_{11} \\ V_{21} \\ V_{31} \\ V_{41} \end{bmatrix} = \begin{bmatrix} \tau_{x} \\ \tau_{x} \\ \tau_{y} \\ \tau_{y} \end{bmatrix}$$
(12)

where $M \in \mathbb{R}^{4 \times 4}$, $V \in \mathbb{R}^{4 \times 1}$. The specific elements of matrices M and V are in Appendix 1.

2.2. State-space equations of the pendulum SR

The dynamic equation (12) is a coupled nonlinear system. It is challenging to construct the motion control system of type XK-I described by Eq. (12). According to the method of [4, 8, 20], the XK-I SR's rolling and steering motion control can be separated; that is, the driving force produced by a group of motors makes the XK-I move in a straight line. If the XK-I type needs steering motion during linear motion, it is only necessary to input control signals to another group of motors. Therefore, the SR's linear motion is driven by the torque τ_x output by one set of motors. In contrast, the torque τ_y output by another set of motors derives the steering movement. This decoupling approach converts the nonholonomic system into two decoupled subsystems, simplifies the SR dynamics equations, and improves the maneuverability of the system.

Besides, in the XK-I type's actual movement process, the pendulum's swing angle is slight, and the Taylor series is utilized to linearize the sine and cosine functions of α and β .

When the SR moves in a straight line, only $u_1 = \tau_x$ is the torque provided by the linear motor. Therefore, the pendulum's swinging angle is only α , and the spherical shell's rolling angle is only θ . Then, the state variable of the SR's linear motion is $\mathbf{x}_1 = [q_1, \dot{q}_1, q_2, \dot{q}_2]^T$. Since the pendulum swing angle α is small during the XK-I spherical robot's actual movement, $\sin \alpha$ and $\cos \alpha$ are described with their corresponding Taylor series. Since for small swinging angles, we have $\sin \alpha = \alpha$ and $\cos \alpha = 1$, the state-space expression of the SR's linear motion can be attained as

$$\begin{cases} \dot{x}_1 = A_1 x_1 + B_1 u_1 \\ y_1 = C_1 x_1 + D_1 u_1 \end{cases}$$
(13)

where $A_1 \in \mathbb{R}^{4 \times 4}$, $B_1 \in \mathbb{R}^{4 \times 1}$, $C_1 \in \mathbb{R}^{4 \times 4}$, and $D_1 = [\mathbf{0}_{4 \times 1}]$. The specific elements of matrices A_1 , B_1 and C_1 are in Appendix 2.



Figure 2. Adaptive control system of spherical robot.

In the same way, when the spherical robot turns and moves, only $u_2 = \tau_y$ is the torque provided by the motor. Therefore, the pendulum's swing angle is only β , and the spherical shell's rolling angle is only φ . Since the pendulum swing angle β is small during the actual movement of the XK-I SR, sin β and cos β can be represented with their corresponding Taylor series. The state variable of the SR turning motion is $\mathbf{x}_2 = [q_3, \dot{q}_3, q_4, \dot{q}_4]^T$. Then, the SR turning motion's state-space equations are

$$\begin{cases} \dot{x}_2 = A_2 x_2 + B_2 u_2 \\ y_2 = C_2 x_2 + D_2 u_2 \end{cases}$$
(14)

where $A_2 \in \mathbb{R}^{4 \times 4}$, $B_2 \in \mathbb{R}^{4 \times 1}$, $C_2 \in \mathbb{R}^{4 \times 4}$, and $D_2 = [\mathbf{0}_{4 \times 1}]$. The specific elements of matrices A_2 , B_2 , and C_2 are in Appendix 3.

3. Design of the XK-I adaptive motion controller

The design block diagram of the model reference adaptive controller of the XK-I SR is presented in Fig. 2. The controller mainly includes a reference model, adaptive regulator, and parameter debugger. Therefore, the XK-I SR controller's reference model is first constructed according to the state space equation and the LQR. Here, we focus on analyzing the XK-I SR's linear motion control, and the steering motion can be obtained similarly. In the XK-I SR's motion control, we give the desired motion state x'_1 , such as the angular velocity and angle of the spherical shell and the pendulum. It is hoped that the SR output is the same as the expected value at each moment. However, the actual process is imperfect, and there will be errors in the control at every moment. When only the error at the current moment is utilized in the controller design, the steady-state error is inevitable, degrading the control system's performance. The desired system performance can be attained by minimizing the accumulated state errors of the XK-I SR in the entire movement time. Then, the controller is designed to minimize the cumulative error when the SR's motion state reaches the desired value at a particular moment. Therefore, the quadratic performance cost function of the error state and control signal u_1 or u_2 is determined to minimize the following quadratic performance index

$$J = \frac{1}{2} \int_0^\infty \left(\boldsymbol{e}_1^T \boldsymbol{\mathcal{Q}}_1 \boldsymbol{e}_1 + R \boldsymbol{u}_1^2 \right) dt \to \min$$
 (15)

where $e_1 \in \mathbb{R}^{4\times 1}$ is the error vector, $e_1 = x'_1 - x_1$, and x'_1 describes the given state vector, $Q_1 \in \mathbb{R}^{4\times 4}$ is a positive definite matrix and *R* describes a weighting constant. The control law that minimizes the performance index Eq. (15) is

$$u_1 = R^{-1} \boldsymbol{B}_1^T \boldsymbol{P} \boldsymbol{e}_1 = \boldsymbol{K} \boldsymbol{e}_1 \tag{16}$$

where *P* is the Riccati equation's positive definite symmetrical solution, then, $A_1^T P + PA_1 + Q_1 - PB_1R^{-1}B_1^T P = 0$, *K* is the control parameter of the optimal controller. Therefore, the reference model's state-space equations are designed as

$$\begin{cases} \dot{\mathbf{x}}_{1r} = \mathbf{A}_1 \mathbf{x}_{1r} + \mathbf{B}_1 u_1 \\ \mathbf{y}_{1r} = \mathbf{C}_1 \mathbf{x}_{1r} \end{cases}$$
(17)

Bringing Eq. (16) into the state equation (17) of the reference model, and assume that the state x_1 can be consistent with the state x_{1r} of the reference model, we get,

$$\begin{cases} \dot{x}_{1r} = A_{1r} x_{1r} + B_{1r} x_1' \\ y_{1r} = C_1 x_{1r} \end{cases}$$
(18)

where $A_{1r} = (A_1 - B_1K)$, $B_{1r} = B_1K$. In practice, it is difficult for x_{1r} to be consistent with x_1 , then the error between the reference model output x_{1r} and the state vector x_1 is $e_{1r} = x_{1r} - x_1$. Combining Eqs. (13) and (18), we can get,

$$\dot{\boldsymbol{e}}_{1r} = \dot{\boldsymbol{x}}_{1r} - \dot{\boldsymbol{x}}_{1} = \boldsymbol{A}_{1r} \boldsymbol{e}_{1r} + (\boldsymbol{A}_{1r} - \boldsymbol{A}_{1}) \boldsymbol{x}_{1} + \boldsymbol{B}_{1r} \boldsymbol{x}_{1}' - \boldsymbol{B}_{1} \boldsymbol{u}_{1}$$
(19)

The goal is to make x_{1r} approach x_1 , then $e_{1r} = 0$. Thus, u_1 can be calculated by Eq. (19),

$$u_{1} = (\boldsymbol{B}_{1}^{T}\boldsymbol{B}_{1})^{-1} \boldsymbol{B}_{1}^{T} (\boldsymbol{A}_{1r} - \boldsymbol{A}_{1}) \boldsymbol{x}_{1} + (\boldsymbol{B}_{1}^{T}\boldsymbol{B}_{1})^{-1} \boldsymbol{B}_{1}^{T}\boldsymbol{B}_{1r} \boldsymbol{x}_{1}' = \boldsymbol{\rho} \boldsymbol{x}_{1} + \boldsymbol{\eta} \boldsymbol{x}_{1}'$$
(20)

Thus, Eq. (16) can be rewritten as Eq. (20). Since the reference model is an ideal control system, ρ and η are ideal tuning parameters, and Eq. (20) is a desired control law. Additionally, since the error term is not included in the control law (20), it is challenging to eliminate the error by adjusting the parameters. Therefore, the actual control law of the XK-I robot's linear motion under complex road conditions is considered as follows:

$$\hat{u}_1 = \hat{\rho} \boldsymbol{x}_1 + \hat{\eta} \boldsymbol{x}_1' \tag{21}$$

Multiplying Eq. (20) by matrix B_1 gives

$$\boldsymbol{B}_{1}\boldsymbol{u}_{1} = (\boldsymbol{A}_{1r} - \boldsymbol{A}_{1})\,\boldsymbol{x}_{1} + \boldsymbol{B}_{1r}\boldsymbol{x}_{1}' = \boldsymbol{B}_{1}\left(\boldsymbol{\rho}\boldsymbol{x}_{1} + \boldsymbol{\eta}\boldsymbol{x}_{1}'\right) \tag{22}$$

Substitute formula (21) into deviation Eq. (19), extracting matrix B_1 , and combining formula (20), we can get

$$\dot{\boldsymbol{e}}_{1r} = \boldsymbol{A}_{1r}\boldsymbol{e}_{1r} + \boldsymbol{B}_1\left(\tilde{\boldsymbol{\rho}}\boldsymbol{x}_1 + \tilde{\boldsymbol{\eta}}\boldsymbol{x}_1'\right) \tag{23}$$

where $\tilde{\rho}$ and $\tilde{\eta}$ are the differences between the ideal and actual adjustment coefficients; that is, $\tilde{\rho} = \rho - \hat{\rho}$, $\tilde{\eta} = \eta - \hat{\eta}$. In order to obtain the actual control law, the following Lyapunov function candidate is defined in terms of the errors e_{1r} and $\tilde{\rho}$, $\tilde{\eta}$ are

$$\boldsymbol{V} = \boldsymbol{e}_{1r}^{T} \boldsymbol{H} \boldsymbol{e}_{1r} + |\boldsymbol{B}_{1}| \left(\frac{1}{2} \tilde{\boldsymbol{\rho}}^{T} \boldsymbol{N}_{1} \tilde{\boldsymbol{\rho}} + \tilde{\boldsymbol{\eta}}^{T} \boldsymbol{N}_{2} \tilde{\boldsymbol{\eta}}\right)$$
(24)

where N_1 , N_2 are diagonal matrices, where their diagonal elements are positive. The matrix H satisfies the following relationship: $A_{1r}^T H + HA_{1r} = -I$. Differentiating Eq. (24) gives

$$\dot{\boldsymbol{V}} = \boldsymbol{e}_{1r}^{T} \left(\boldsymbol{A}_{1r}^{T} \boldsymbol{H} + \boldsymbol{H} \boldsymbol{A}_{1r} \right) \boldsymbol{e}_{1r} + 2\boldsymbol{e}_{1r}^{T} \boldsymbol{H} \boldsymbol{B} \left(\tilde{\boldsymbol{\rho}} \boldsymbol{x}_{1} + \tilde{\boldsymbol{\eta}} \boldsymbol{x}_{1}^{'} \right) + |\boldsymbol{B}_{1}| \left(\tilde{\boldsymbol{\rho}}^{T} \boldsymbol{N}_{1} \dot{\tilde{\boldsymbol{\rho}}} + \tilde{\boldsymbol{\eta}}^{T} \boldsymbol{N}_{2} \dot{\tilde{\boldsymbol{\eta}}} \right)$$
(25)

Since the first term in Eq. (25) is negative, the following equality should be satisfied to ensure $\dot{V} \leq 0$,

$$2\boldsymbol{e}_{1r}^{T}\boldsymbol{H}\boldsymbol{B}\big(\tilde{\boldsymbol{\rho}}\boldsymbol{x}_{1}+\tilde{\boldsymbol{\eta}}\boldsymbol{x}_{1}^{\prime}\big)+|\boldsymbol{B}_{1}|\left(\tilde{\boldsymbol{\rho}}^{T}\boldsymbol{N}_{1}\dot{\boldsymbol{\dot{\rho}}}+\tilde{\boldsymbol{\eta}}^{T}\boldsymbol{N}_{2}\dot{\boldsymbol{\dot{\eta}}}\right)=0$$
(26)

Equation (26) has $\tilde{\rho}$, $\tilde{\eta}$ variables. If $\tilde{\rho}$ matches with $\tilde{\eta}$, then Eq. (26) has a solution. Therefore, assuming $N_1 = N_2 = N$, $-\rho = \eta$, and $-\hat{\rho} = \hat{\eta}$ we can get $-\tilde{\rho} = \tilde{\eta}$, simplifies Eq. (26) as

$$-2\boldsymbol{e}_{1r}^{T}\boldsymbol{H}\boldsymbol{B}\tilde{\boldsymbol{\rho}}\boldsymbol{e}_{1}+2\left|\boldsymbol{B}_{1}\right|\tilde{\boldsymbol{\rho}}^{T}\boldsymbol{N}\dot{\tilde{\boldsymbol{\rho}}}=0$$
(27)

Parameter	Value	Parameter	Value
$\overline{m_1}$	0.6 kg	I_1	$0.009 \text{ kg} \cdot \text{m}^2$
m_2	0.51 kg	I_2	$0.0017 \text{ kg} \cdot \text{m}^2$
R	0.15 m	l	0.1 m
g	9.8 N/m		

Table I. Robot parameters.



Figure 3. XK-I spherical robot prototype.

When $-\dot{\hat{\rho}} = \dot{\tilde{\rho}}$, the control law and coefficient adjustment rule are obtained as

$$\hat{u}_1 = \hat{\rho} \boldsymbol{e}_1 \tag{28}$$

$$\dot{\hat{\boldsymbol{\rho}}} = -\operatorname{sgn}\boldsymbol{B}_1 \cdot \boldsymbol{N}^{-1}\boldsymbol{e}_{1r}^T \boldsymbol{H} \boldsymbol{e}_1$$
⁽²⁹⁾

4. Experimental analysis and discussion

In order to evaluate the performance of the designed adaptive motion controller, simulation and real experiments of the XK-I SR are performed. The simulation experiments in this study are divided into two groups: (1) The ideal situation without disturbance and noise, in which the designed controller speed is verified and (2) random interference noise is added to the feedback loop to evaluate the antiinterference capability of the constructed controller. In order to evaluate the efficiency of the MRAC, the simulation experiments are compared with LQR, Fuzzy-PID, PSO-ADRC, and HSMC.

4.1. Simulation results

Table I shows the parameter settings in the simulation process based on the XK-I SR's physical prototype (as presented in Fig. 3). During the simulation, the linear and steering motions are carried out independently; that is, the motor groups M1 and M2 control the linear and steering motions, respectively.

EC NL			Ε				
	NL	NM	NS	ZE	PS	РМ	PL
NL	PL	PL	PM	PM	PS	PS	ZE
NM	PL	PL	PM	PM	PS	ZE	ZE
NS	PM	PM	PM	PS	ZE	NS	NM
ZE	PM	PS	PS	ZE	NS	NM	NM
PS	PS	PS	ZE	NS	NS	NM	NM
PM	ZE	ZE	NS	NM	NM	NM	NL
PL	ZE	NS	NS	NM	NM	NL	NL

Table II. Fuzzy rules for Kp parameters in Fuzzy-PID.

Table III. Fuzzy rules for Ki parameters in Fuzzy-PID.

EC NL							
	NL	NM	NS	ZE	PS	РМ	PL
NL	NL	NL	NL	NM	NM	ZE	ZE
NM	NL	NL	NM	NM	NS	ZE	ZE
NS	NM	NM	NS	NS	ZE	PS	PS
ZE	NM	NS	NS	ZE	PS	PS	PM
PS	NS	NS	ZE	PS	PS	PM	PM
PM	ZE	ZE	PS	PM	PM	PL	PL
PL	ZE	ZE	PS	PM	PL	PL	PL

EC	E								
	NL	NM	NS	ZE	PS	PM	PL		
NL	PS	PS	ZE	ZE	ZE	PL	PL		
NM	NS	NS	NS	NS	ZE	NS	PM		
NS	NL	NL	NM	NS	ZE	PS	PM		
ZE	NL	NM	NM	NS	ZE	PS	PM		
PS	NL	NM	NS	NS	ZE	PS	PS		
PM	NM	NS	NS	NS	ZE	PS	PS		
PL	PS	ZE	ZE	ZE	ZE	PL	PL		

Table IV. Fuzzy rules for Kd parameters in Fuzzy-PID.

Besides, the rotational speed of the XK-I SR's linear and steering motions is set to 3 rad/s, and random signals are selected as the interference source, where the interference amplitude is less than 1 rad/s.

In the simulations, when the LQR controller is utilized to control the motion effect, the cumulative amount of the state change of the system should be as small as possible during the change of the state variable of the robot from the initial value to the expected value. In this way, the influence of the disturbance caused by the dynamic change in the robot's state can be reduced. Therefore, Eq. (15) is adopted as the system's objective function.

When designing the Fuzzy-PID controller, the input error is quantified in the range (-6,6). Further, the error is fuzzified to determine the fuzzy subset as {NL, NM, NS, ZE, PS, PM, PL}. Besides, the triangular membership function is selected. The specific fuzzy rules are presented in Tables II–IV.



Figure 4. Linear motion simulation of the SR: (a) no interference environment; (b) interference environment.



Figure 5. Errors of each controller and reference signal in linear motion of the SR: (a) no interference environment; (b) interference environment.

The PSO-ADRC controller adopts the method provided in ref. [20]. In the particle swarm algorithm, both the population size and the number of iterations are set to 50 so that the overall number of iterations of the ADRC controller reaches 2500, and the random generation of parameters ranges from 0.0001 to 100000 values. The controller tuning parameters mainly select the velocity factor r and the system integration time h in the tracking differentiator (TD) in ADRC, and the compensation coefficient b_0 and the observer parameters b_{01} , b_{02} , b_{03} in the extended state observer (ESO) can be considered a coupling of two second-order subsystems. The HSMC controller selects the relationship between the spherical shell and the pendulum in the state-space equations as the target-controlled object and calculates the difference between the output velocity and acceleration of the spherical shell and the ideal signal.

During simulation, the ideal model's output speed is selected as the reference speed signal (3 rad/s). The linear motion simulation results are presented in Figs. 4 and 5. As shown in Figs. 4 and 5, the presented MRAC strategy makes the output velocity approach the reference signal quickly under ideal and interference conditions. In terms of the overall speed, in the ideal case, we have MRAC>PSO-ADRC>Fuzzy-PID>LQR>HSMC; in the case of interference, we have MRAC>PSO-ADRC>Fuzzy-PID>LQR. In terms of error volatility, Fuzzy-PID, LQR, and HSMC have noticeable fluctuations with a specific magnitude under ideal conditions; in the case of interference, in the error change at 0.5 s, $e_{MRAC} < e_{PSO-ADRC} < e_{Fuzzy-PID} < e_{HSMC} < e_{LQR}$. Fuzzy-PID, LQR, and HSMC have evident error fluctuations in the next few seconds.

Similarly, the steering motion simulation results can be obtained, as shown in Figs. 6 and 7. Under ideal conditions, in terms of fast performance, we have MRAC>PSO-ADRC>Fuzzy-PID>LQR>SMC. After adding interference, the five controllers all have significant fluctuations in the initial period at 0.5 s, while the MRAC controller fluctuation is close to the impulse signal. It can be seen that the reference model can be quickly tracked even under the action of disturbance, which can efficiently counter the disturbance and restore the system's stability quickly.



Figure 6. Steering motion simulation of SR: (a) interference-free environment; (b) interference environment.



Figure 7. Errors between the controllers and the reference signal in the spherical robot's linear motion: (*a*) *no interference environment;* (*b*) *interference environment.*



Figure 8. The spherical robot's linear motion control under the smooth ground condition.

4.2. Linear motion experiment

Further, real experiments of linear motion control are performed on smooth and uneven ground. The motor velocity's expected values of the SR are set as 50 rps, 100 rps, and 150 rps, respectively. The experimental results are presented in Figs. 8 and 9, and the analysis data are presented in Table V. Figures 10 and 11 show the process of using the MRAC controller to control the XK-I linear motion under smooth and uneven ground conditions.

Experimental conditions	Expected value		LQR	Fuzzy-PID	PSO-ADRC	HSMC	MRAC
Smooth ground	50 rps	Error	3.75%	0.38%	2.50%	0.87%	0.13%
		Mean	48.125	50.188	48.750	49.563	50.063
		Variance	0.250	0.696	15.133	0.930	0.196
	100 rps	Error	5.56%	1.19%	6.19%	1.13%	1.94%
		Mean	94.438	98.813	93.813	101.125	98.063
		Variance	85.995	2.029	66.296	2.117	0.195
	150 rps	Error	13.54%	0.54%	4.37%	1.08%	0.12%
		Mean	129.688	149.188	143.438	148.375	149.813
		Variance	573.162	2.696	219.595	2.917	0.296
Uneven ground	50 rps	Error	30.25%	11.50%	12.00%	16.37%	19.25%
		Mean	34.875	44.25	44.000	41.813	40.375
		Variance	421.717	135.133	172.667	143.229	145.717
	100 rps	Error	44.81%	6.38%	13.88%	10.13%	23.13%
		Mean	55.188	93.625	86.125	89.875	76.875
		Variance	694.963	418.917	765.717	291.45	591.583
	150 rps	Error	51.58%	7.08%	3.04%	6.5%	8.29%
	-	Mean	72.625	139.375	145.437	140.25	137.563
		Variance	1517.45	290.65	263.863	308.467	374.529

 Table V. Control error and stability comparison table of four controllers under different conditions.



Figure 9. The spherical robot's linear motion control under the uneven ground condition.



Figure 10. Process diagram of the XK-I SR moving on smooth ground.

Figure 8 presents the SR test results on smooth ground. Due to the error of modeling and model decoupling, the system parameters may change during the SR's actual motion. Figure 8 shows that Fuzzy-PID, MRAC, and HSMC can control the movement speed to reach the desired value within 1s,



Figure 11. Process diagram of the XK-I SR moving on uneven ground.

while the transient response efficiency of the LQR and PSO-ADRC controllers is low. Among the Fuzzy-PID, MRAC, and HSMC controllers, the Fuzzy-PID controller leads to a considerable overshoot in the transient response. The transient response fluctuations of the Fuzzy-PID controller are more significant than the MRAC. The HSMC controller always has evident static error problems and is slower than the MRAC and Fuzzy-PID controllers. Besides, Table V indicates that the movement speed variance provided by the MRAC is 0.196, 0.195, and 0.296, respectively, which is the smallest variance compared to the other four controllers. These features demonstrate the good control effect of the MRAC controller.

Figure 9 presents the SR test results on the bumpy ground. Under strong interference, all five controllers exhibit significant fluctuations. In terms of speed, MRAC, Fuzzy-PID, and HSMC controllers all provide good results. Besides, the designed MRAC's anti-interference capability is at a medium level, which is stronger than the LQR controller and the same as the HSMC controller but weaker than the PSO-ADRC and Fuzzy-PID controllers.

In summary, in the simulation and real experiments, the MRAC controller can quickly approach the control motor's expected value of the SR and quickly restore stability in the presence of disturbances. The simulation and smooth ground test results demonstrate that the MRAC still has good control ability under motion disturbances.

The data in Table V are divided into two cases: smooth ground and uneven ground. Under smooth ground, the mean value of MRAC is closest to the target value. The error is within 2 rps. The control effect of other controllers is quite different from the set value. The control signal error follows MRAC<Fuzzy-PID<HSMC<PSO-ADRC<LQR. In terms of variance, the volatility of MRAC is also the smallest. The fluctuation of the control signal follows MRAC<Fuzzy-PID<HSMC<PSO-ADRC<LQR. The overall reflection of the fast performance of MRAC is the best.

Under uneven ground, there is external interference. As presented in Table V, both the mean and variance of MRAC are in the middle of several controllers. Figure 8 intuitively shows that the fast performance of MRAC is very prominent, while its anti-interference performance is not optimal. Besides, the noise immunity performances of Fuzzy-PID, HSMC, and PSO-ADRC are almost the same.

The ideal and interference environments in Table V reflect the controller's fast and anti-interference performance, respectively. The data reflect that the designed MRAC has excellent fast performance, while the anti-interference is insufficient. In the real environment, various disturbances, such as the changes in the external environment, will inevitably affect the SRs' motion control performance. In the future, combined with the MRAC, a motion control strategy for SRs that can effectively suppress external disturbances, such as a disturbance observer-based control method [33], will be studied. In this method, the difference between the actual object and the nominal model induced by external disturbances and model parameter variations is equivalent to the control input, and equivalent compensation is incorporated into the controller to eliminate the disturbances.

4.3. Rolling of the XK-I SR over curvilinear trajectories

In order to verify the practical application of the presented control approach, an experiment was performed in which the MRAC controlled the XK-I SR to move along a curve on smooth ground. The trajectory diagram of moving along the curve is presented in Fig. 12, and the actual scene diagrams of the XK-I SR moving along the curve are shown in Figs. 13 and 14. Figure 12 shows a particular deviation when the MRAC controls the XK-I SR to move along the curve, indicating that it cannot accurately track



Figure 12. Experimental results of rolling over curvilinear trajectories.



Figure 13. The actual scene of the XK-I SR moving along a circular trajectory.

the curve movement. During the MRAC design, controlling the angular velocity and angle of the XK-I SR's pendulum swing and spherical shell rolling is considered while ignoring its position and attitude control, deviating from the control of the XK-I along the curve. In the future, the expected value in the MRAC can be set to the trajectory coordinate value moving along the curve to reconstruct the control law and realize the XK-I SR's path following control.

5. Conclusions

The current work presented a Lyapunov-based MRAC control algorithm. The SR's omnidirectional dynamic model is established via the Euler–Lagrange method by taking the pendulum's swing angle and the spherical shell's rolling angle as the generalized coordinates. According to the steering principle of the car during driving, the XK-I SR's dynamic model is decoupled into linear and steering models. Besides, MRAC is utilized to design the SR's adaptive motion control law, and a Lyapunov-based adaptation mechanism for the controller parameters is designed to overcome the poor control effect induced



Figure 14. The actual scene of the XK-I SR moving along the trajectory.

by the SR's inaccurate dynamic model and parameter uncertainty. Finally, to evaluate the effect of the designed motion controller, the simulation and real experiments are compared with LQR, Fuzzy-PID, PSO-ADRC, and HSMC. Experiments indicate that the designed controller can be successfully applied to the SR control system. In the case of interference, it can quickly restore stability. Especially when the system parameters are uncertain, the SR can still be well controlled to achieve the expected effect after decoupling the SR. In the future, the SR's control system should be further studied, mainly to solve the anti-interference ability problem and trajectory tracking of the MRAC controller.

Author contributions. All authors proposed the research. Jianwen Huo conceived and designed the study. Rui Lin conducted data gathering and performed statistical analyses. Jianwen Huo and Rui Lin wrote the article.

Financial support. This research was funded by the Research Fund of Southwest University of Science and Technology (No. 22zx7109).

Conflicts of interest. The authors declare that they have no conflicts of interest.

Ethical approval. None.

Acknowledgement. The authors would like to thank the financial support from the Research Fund of Southwest University of Science and Technology.

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Appendix 1

 $m_1 = (m_1 + m_2)r^2 + I_1 + I_2 + 2m_2l^2 \sin\varphi \cos\varphi \cos\alpha \sin\beta \cos\beta + m_2l^2(\sin\alpha^2 + \sin\varphi^2 \sin\beta^2 + \cos\varphi^2 \cos\alpha^2 \cos\beta^2);$ $m_2 = I_2 + m_2l^2(\sin\alpha^2 + \cos\alpha^2 \cos\beta^2);$

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m_3 = (m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2(\cos\varphi^2\sin\beta^2 + \sin\theta^2\sin\alpha^2 - 2\sin\varphi\cos\varphi\cos\alpha\sin\beta\cos\beta)
+\cos \alpha^2 \sin \varphi^2 \cos \beta^2 + \cos \theta^2 \sin \varphi^2 \sin \beta^2 + 2\sin \theta \cos \theta \sin \alpha \cos \alpha \cos \beta \cos \varphi
+\cos\theta^2\cos\alpha^2\cos\varphi^2\cos\beta^2 + 2\sin\theta\cos\theta\sin\varphi\sin\alpha\sin\beta + 2\cos\theta^2\sin\varphi\cos\varphi\cos\alpha\sin\beta\cos\beta;
m_4 = I_2 + m_2 l^2 (\sin \beta^2 + \cos \alpha^2 \cos \beta^2);
a = -I_2 \cos \varphi + m_2 lr \sin \varphi \sin \alpha + m_2 l^2 (-\sin \varphi \cos \alpha \sin \beta \cos \beta - \cos \varphi \cos \alpha^2 \cos \beta^2 - \cos \varphi \sin \alpha^2);
b = m_2 l^2 (\sin \theta \sin \varphi^2 \sin \beta \cos \beta \cos \alpha - \sin \theta \sin \varphi \cos \varphi \sin \beta^2 - \cos \theta \sin \varphi \sin \alpha \cos \alpha \cos \beta
+\cos\theta\cos\varphi\sin\alpha\sin\beta-\sin\theta\cos\varphi^2\cos\alpha\sin\beta\cos\beta+\sin\theta\sin\varphi\cos\varphi\cos\alpha^2\cos\beta^2);
c = m_2 lr(\sin\varphi\sin\beta + \cos\varphi\cos\alpha\cos\beta) + m_2 l^2(+\sin\varphi\sin\alpha\cos\alpha\cos\beta - \cos\varphi\sin\beta\sin\alpha);
d = -I_2 \sin \theta \sin \varphi + m_2 lr(-\cos \theta \cos \alpha \cos \beta - \sin \theta \cos \varphi \sin \alpha) + m_2 l^2(-\cos \theta \sin \alpha \sin \beta)
-\sin\theta\sin\varphi\sin\alpha^2 + \sin\theta\cos\varphi\cos\alpha\sin\beta\cos\beta - \sin\theta\sin\varphi\cos\alpha^2\cos\beta^2;
e = m_2 l^2 (\sin \alpha \sin \beta):
f = -I_2 \cos \theta + m_2 lr(-\sin \theta \cos \varphi \sin \beta - \sin \theta \sin \varphi \cos \alpha \cos \beta) + m_2 l^2(-\cos \theta \sin \beta^2)
 -\cos\theta\cos\alpha^2\cos\beta^2 - \sin\theta\sin\varphi\sin\alpha\sin\beta - \sin\theta\sin\varphi\sin\alpha\cos\alpha\cos\beta.);
V_{11} = m_2 r l (-\dot{\alpha} (\sin \varphi \sin \alpha)' + \dot{\beta} (\sin \varphi \sin \beta + \cos \varphi \cos \alpha \cos \beta)')
      +m_2 l^2 (\dot{\alpha}(-\sin\varphi\cos\alpha\sin\beta\cos\beta-\cos\varphi\cos\alpha^2\cos\beta^2-\cos\varphi\sin\alpha^2)'
      +\dot{\beta}(-\cos\varphi\sin\alpha\sin\beta + \sin\varphi\sin\alpha\cos\alpha\cos\beta)' + \dot{\theta}(2\sin\varphi\cos\varphi\cos\alpha\sin\beta\cos\beta)
      +\cos \varphi^2 \cos \alpha^2 \cos \beta^2 + \sin \varphi^2 \sin \beta^2 + \sin \alpha^2)' + \dot{\varphi}(-\sin \theta \sin \varphi \cos \varphi \sin \beta^2)
      +\sin\theta\sin\varphi^2\cos\alpha\sin\beta\cos\beta + \cos\theta\cos\varphi\sin\alpha\sin\beta - \cos\theta\sin\varphi\sin\alpha\cos\alpha\cos\beta
      -\sin\theta\cos\varphi^2\cos\alpha\sin\beta\cos\beta + \sin\theta\sin\varphi\cos\varphi\cos\alpha^2\cos\beta^2)'
      +I_2(\dot{\alpha}\dot{\phi}\sin\varphi+\dot{\beta}\dot{\phi}\sin\theta-\dot{\alpha}\dot{\phi}\cos\theta\sin\varphi)+m_2(Z_1\cdot(-\dot{\phi}l\sin\theta\sin\varphi\sin\beta+\dot{\phi}l\cos\theta\sin\alpha))
      -\dot{\varphi}l\sin\theta\cos\varphi\cos\alpha\cos\beta + Z_2 \cdot (-\dot{\alpha}l\sin\theta\cos\alpha\cos\beta + \dot{\alpha}l\cos\theta\cos\varphi\sin\alpha)
      +\dot{\beta}l\cos\theta\cos\varphi\sin\beta-\dot{\beta}l\cos\theta\sin\varphi\cos\alpha\cos\beta+\dot{\theta}l\sin\theta\sin\varphi\sin\beta
      -\dot{\theta}l\cos\theta\sin\alpha + \dot{\theta}l\sin\theta\cos\varphi\cos\alpha\cos\beta + Z_3 \cdot (-\dot{\alpha}l\sin\theta\cos\varphi\sin\alpha)
      -\dot{\alpha}l\cos\theta\cos\alpha\cos\beta + \dot{\theta}l\cos\theta\sin\phi\sin\beta + \dot{\theta}l\sin\theta\sin\alpha + \dot{\theta}l\cos\theta\cos\phi\cos\alpha\cos\beta
      -\dot{\beta}l\sin\theta\cos\varphi\sin\beta+\dot{\beta}l\sin\varphi\sin\theta\cos\alpha\cos\beta;
V_{21} = I_2(\dot{\theta}\dot{\varphi}\sin\varphi - \dot{\theta}\dot{\varphi}\cos\theta\sin\varphi - \dot{\varphi}^2\sin\theta\cos\varphi)
      +m_2 r l(\dot{\theta}(\sin\varphi\sin\alpha)' + \dot{\varphi}(-\cos\theta\cos\alpha\cos\beta - \sin\theta\cos\varphi\sin\alpha)')
      +m_2 l^2 (\dot{\alpha} (\sin \alpha^2 + \cos \alpha^2 \cos \beta^2)' + \dot{\beta} (\sin \alpha \sin \beta)'
      +\dot{\theta}(-\sin\varphi\cos\alpha\sin\beta\cos\beta-\cos\varphi\sin\alpha^2-\cos\varphi\cos\alpha^2\cos\beta^2)'
      +\dot{\varphi}(-\cos\theta\sin\alpha\sin\beta-\sin\theta\sin\varphi\sin\alpha^2+\sin\theta\cos\varphi\cos\alpha\sin\beta\cos\beta)
      -\sin\theta\sin\varphi\cos\alpha^2\cos\beta
      +m_2(Z_1 \cdot (-\dot{\alpha} l\sin\varphi\cos\alpha + \dot{\beta} l\cos\varphi\sin\alpha\cos\beta + \dot{\varphi} l\sin\theta\cos\alpha - \dot{\varphi} l\cos\theta\cos\varphi\sin\alpha\cos\beta)
      +Z_2 \cdot (-\dot{\alpha} l\cos\theta\sin\alpha\cos\beta + \dot{\alpha} l\sin\theta\cos\varphi\cos\alpha + \dot{\beta} l\sin\theta\sin\varphi\sin\alpha\cos\beta)
      -\dot{\theta}l\sin\theta\cos\alpha + \dot{\theta}l\cos\theta\cos\varphi\sin\alpha\cos\beta
      +Z_3 \cdot (\dot{\alpha} l \cos \theta \cos \varphi \cos \alpha + \dot{\alpha} l \sin \theta \sin \alpha \cos \beta)
      -\dot{\theta}l\cos\theta\cos\alpha - \dot{\theta}l\sin\theta\cos\varphi\sin\alpha\cos\beta + \dot{\beta}l\sin\varphi\cos\theta\sin\alpha\cos\beta
      -\dot{\varphi}l\sin\varphi\sin\alpha\cos\beta) -m_2gl\sin\alpha\cos\beta;
V_{31} = I_2(-\dot{\alpha}\dot{\theta}\cos\theta\sin\varphi - 2\dot{\alpha}\dot{\varphi}\sin\theta\cos\varphi + \dot{\beta}\dot{\theta}\sin\theta + \dot{\theta}\dot{\alpha}\sin\varphi)
      +m_2 r l(\dot{\alpha}(-\cos\theta\cos\alpha\cos\beta-\sin\theta\cos\varphi\sin\alpha)')
      +\beta(-\sin\theta\cos\varphi\sin\beta+\sin\theta\sin\varphi\cos\alpha\cos\beta)')
      +m_2l^2(\dot{\alpha}(\sin\theta\cos\varphi\cos\alpha\sin\beta\cos\beta-\cos\theta\sin\alpha\sin\beta-\sin\theta\sin\varphi\sin\alpha^2
      -\sin\theta\sin\varphi\cos\alpha^2\cos\beta^2
      +\dot{\beta}(-\cos\theta\sin\beta^2-\cos\theta\cos\alpha^2\cos\beta^2-\sin\theta\sin\varphi\sin\alpha\sin\beta-\sin\theta\cos\varphi\sin\alpha\cos\beta)
      +\dot{\varphi}(\cos\theta^2\sin\varphi^2\sin\beta^2+2\sin\theta\cos\theta\sin\varphi\sin\alpha\sin\beta+2\cos\theta^2\sin\varphi\cos\varphi\cos\alpha\sin\beta\cos\beta)
      +\sin\theta^2\sin\alpha^2+2\sin\theta\cos\theta\cos\phi\sin\alpha\cos\beta+\cos\theta^2\cos\phi^2\cos\alpha^2\cos\beta^2+\cos\phi^2\sin\beta^2
      -2\sin\varphi\cos\varphi\cos\alpha\sin\beta\cos\beta+\sin\varphi^2\cos\alpha^2\cos\beta^2)'
      +\dot{\theta}(-\sin\theta\sin\varphi\cos\varphi\sin\beta^2+\cos\theta\cos\varphi\sin\alpha\sin\beta-\sin\theta\cos\varphi^2\cos\alpha\sin\beta\cos\beta)
      +\sin\theta\sin\varphi^2\cos\alpha\sin\beta\cos\beta-\cos\theta\sin\varphi\sin\alpha\cos\alpha\cos\beta+\sin\theta\sin\varphi\cos\varphi\cos\alpha^2\cos\beta')
      +m_2(Z_1 \cdot (-\dot{\alpha}l\cos\varphi\sin\alpha - \dot{\beta}l\cos\varphi\sin\beta + \dot{\beta}l\sin\varphi\cos\alpha\cos\beta + \dot{\varphi}l\cos\theta\cos\varphi\sin\beta)
      -\dot{\varphi}l\cos\theta\sin\varphi\cos\alpha\cos\beta) + Z_2
      (-\dot{\alpha}l\cos\theta\sin\alpha\cos\beta-\dot{\alpha}l\sin\theta\sin\alpha\sin\varphi-\dot{\beta}l\sin\theta\sin\varphi\sin\beta-\dot{\beta}l\sin\theta\cos\varphi\cos\alpha\cos\beta)
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 $Z_2 = \alpha t \cos \theta \cos \alpha \cos \rho + \alpha t \sin \theta \cos \phi \sin \alpha + \beta t \sin \theta \cos \phi \sin \rho - \beta t \sin \theta \sin \phi \cos \alpha - \dot{\theta} t \cos \theta \sin \phi \sin \beta - \dot{\theta} t \sin \theta \sin \alpha - \dot{\theta} t \cos \theta \cos \phi \cos \alpha \cos \beta - t \dot{\phi};$ $Z_2 = \dot{\alpha} t \cos \theta \cos \phi \sin \alpha - \dot{\alpha} t \sin \theta \cos \alpha \cos \beta + \dot{\theta} t \sin \theta \sin \phi \sin \beta - \dot{\theta} t \cos \theta \sin \alpha$

$$Z_3 = \alpha l \cos \theta \cos \varphi \sin \alpha - \alpha l \sin \theta \cos \alpha \cos \beta + \theta l \sin \theta \sin \varphi \sin \varphi \sin \beta - \theta l \cos \theta \sin \alpha + \theta l \sin \theta \cos \varphi \cos \alpha \cos \beta + \beta l \cos \theta \cos \varphi \sin \beta - \beta l \sin \varphi \cos \theta \cos \alpha \cos \beta - \theta l \cos \varphi \sin \beta + \theta l \sin \varphi \cos \alpha \cos \beta;$$

Appendix 2

$$\begin{split} F_1 &= ((I_2 + m_2 l^2)^3 - (I_2 + m_2 l^2)^2 ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2) + (I_2 + m_2 l^2)m_2^2 l^2 r^2)m_2gl; \\ F_2 &= (l_4 m_2 l_2^2 l^2 r^2 ((m_1 + m_2)r^2 + I_1 + I_2) - ((m_1 + m_2)r^2 + I_1 + I_2)((m_1 + m_2)r^2 + I_1 + I_2)(m_1 + m_2)r^2 \\ &+ I_1 + I_2 + m_2 l^2)(I_2 + m_2 l^2) + ((m_1 + m_2)r^2 + I_1 + I_2)(I_2 + m_2 l^2)^2m_2gl; \\ H_1 &= ((m_1 + m_2)r^2 + I_1 + I_2)(I_2 + m_2 l^2)^3 - ((m_1 + m_2)r^2 + I_1 + I_2)(m_1 + m_2)r^2 \\ &+ I_1 + I_2 + m_2 l^2)(I_2 + m_2 l^2)^2 + ((m_1 + m_2)r^2 + I_1 + I_2)(I_2 + m_2 l^2)m_2^2 l^2 r^2 \\ &+ (I_2 + m_2 l^2)^4 ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2) - (I_2 + m_2 l^2)^4 - 2(I_2 + m_2 l^2)^2m_2^2 l^2 r^2 \\ &+ ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)(I_2 + m_2 l^2)m_2^2 l^2 r^2 - m_2^4 l^4 r^4; \\ F_3 &= (I_2 + m_2 l^2)^2 ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2) - (I_2 + m_2 l^2)^2 - (I_2 + m_2 l^2)m_2^2 l^2 r^2; \\ F_4 &= -((m_1 + m_2)r^2 + I_1 + I_2)(I_2 + m_2 l^2)m_2 lr + (I_2 + m_2 l^2)^2m_2 lr + m_2^3 l^3 r^3 \\ &- (I_2 + m_2 l^2)^2 ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2) + (I_2 + m_2 l^2)^3 + (I_2 + m_2 l^2)m_2^2 l^2 r^2; \end{split}$$

Appendix 3

$$\begin{split} K_1 &= ((I_2 + m_2 l^2)^3 - ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)(I_2 + m_2 l^2)^2 + m_2^2 l^2 r^2 (I_2 + m_2 l^2))m_2 gl; \\ K_2 &= (((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)^2 - ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)m_2^2 l^2 r^2 + ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)m_2^2 l^2 r^2 + ((m_1 + m_2)r^2 + I_1 + I_2 + m_2 l^2)^2 (I_2 + m_2 l^2)m_2; \end{split}$$

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$$\begin{split} &K_3 = (-(m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)^2 + (I_2 + m_2l^2)^3 + (I_2 + m_2l^2)m_2^2l^2r^2 \\ &-((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)m_2lr + (I_2 + m_2l^2)^2m_2lr + m_2^3l^3r^3; \\ &K_4 = -(I_2 + m_2l^2)((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)^2 + ((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)m_2^2l^2r^2 \\ &+((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)^2; \\ &H_2 = ((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)^3 - ((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)^2(I_2 + m_2l^2)^2 \\ &+((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)m_2^2l^2r^2 + ((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)^3 \\ &-(I_2 + m_2l^2)^4 - 2(I_2 + m_2l^2)^2m_2^2l^2r^2 + ((m_1 + m_2)r^2 + I_1 + I_2 + m_2l^2)(I_2 + m_2l^2)^3 \\ &-m_2^4l^4r^4; \end{split}$$

Cite this article: J. Huo, R. Lin and M. Yang (2023). "Design and validation of a novel adaptive motion control for a pendulum spherical robot", Robotica **41**, 2031–2049. https://doi.org/10.1017/S0263574723000280